

(Winter 2007/2008)

1. Consider the 1-DOF system described by the equation of motion,  $4\ddot{x} + 20\dot{x} + 25x = f$ .

- (a) Find the natural frequency  $\omega_n$  and the natural damping ratio  $\xi_n$  of the natural (passive) system ( $f = 0$ ). What type of system is this (oscillatory, overdamped, etc.) ?

Using Section 7.2.2 from the course reader, we can compare this system with  $m\ddot{x} + b\dot{x} + kx = 0$  like in Equation 7.9. Thus:

$$\omega_n = \sqrt{\frac{k}{m}} = \sqrt{\frac{25}{4}} = 2.5$$

$$\xi_n = \frac{b}{2\sqrt{km}} = \frac{20}{2\sqrt{25 \cdot 4}} = 1$$

Since  $\xi_n = 1$ , this system is **critically damped**.

- (b) Design a PD controller that achieves critical damping with a closed-loop stiffness  $k_{CL} = 36$ . In other words, let  $f = -k_v\dot{x} - k_p x$ , and determine the gains  $k_v$  and  $k_p$ . Assume that the desired position is  $x_d = 0$ .

The original system is:

$$4\ddot{x} + 20\dot{x} + 25x = f$$

with input force  $f$ . The controller provides this input, using the formula:

$$f = -k_v\dot{x} - k_p x$$

So, the closed loop equation is:

$$4\ddot{x} + 20\dot{x} + 25x = -k_v\dot{x} - k_p x$$

$$\Rightarrow 4\ddot{x} + (20 + k_v)\dot{x} + (25 + k_p)x = 0$$

This closed loop system behaves just like the natural dissipative system in Section 7.2.2 of the course reader. So, we first compare to Equation 7.9:

$$4\ddot{x} + (20 + k_v)\dot{x} + (25 + k_p)x = m\ddot{x} + b\dot{x} + kx = 0$$

The closed loop stiffness is given by  $k$ , the coefficient of the positional term, so:

$$k = 25 + k_p = k_{CL} = 36$$

For the damping requirement, we need to first figure out how the control gains  $k_p$  and  $k_v$  affect the damping ratio  $\xi$ ; we do this by applying Equation 7.12 to our system:

$$\xi = \frac{b}{2\sqrt{km}}$$

For critical damping:

$$\xi = 1$$

so the coefficient  $b$  of  $\dot{x}$  must satisfy

$$b = 2\sqrt{km} = 2\sqrt{36 \cdot 4} = 24$$

Based on our closed loop equation, we have:

$$b = 20 + k_v$$

so the gains that we need are

$$k_p = 11, \quad k_v = 4$$

and the PD controller is

$$f = -4\dot{x} - 11x$$

- (c) **Assume that the friction model changes from linear ( $20\dot{x}$ ) to Coulomb friction,  $30\text{sign}(\dot{x})$ . Design a control system which uses a non-linear model-based portion with trajectory following to critically damp the system at all times and maintain a closed-loop stiffness of  $k_{CL} = 36$ . In other words, let  $f = \alpha f' + \beta$  and  $f' = \ddot{x}_d - k'_v(\dot{x} - \dot{x}_d) - k'_p(x - x_d)$ . Then, find  $f, \alpha, \beta, f', k'_p$  and  $k'_v$ . Note that  $f$  is an  $m$ -mass control, and  $f'$  is a unit-mass control. Use the definition of error,  $e = x - x_d$ .**

The differential equation for the system is now

$$4\ddot{x} + 30\text{sign}(\dot{x}) + 25x = f$$

In order to linearize it, we apply a force  $f$  of the form

$$f = \alpha f' + \beta$$

where

$$\alpha = 4, \quad \beta = 30\text{sign}(\dot{x}) + 25x$$

For purposes of control, this makes the system look like the unit-mass system:

$$\ddot{x} = f'$$

to which we apply the control

$$f' = \ddot{x}_d - k'_v(\dot{x} - \dot{x}_d) - k'_p(x - x_d)$$

Substituting into our unit-mass system yields the equation

$$\ddot{e} + k'_v\dot{e} + k'_pe = 0$$

where  $e$  is the position error,  $e = x - x_d$ .

Now, we want to choose our gains  $k'_p$  and  $k'_v$  so that we achieve critical damping and the desired closed-loop stiffness.

To look at the closed loop stiffness, we need to consider the controlled system before factoring out the mass:

$$4\ddot{x} + 30\text{sign}(\dot{x}) + 25x = \alpha f' + \beta$$

$$\Rightarrow 4\ddot{x} = \alpha f'$$

$$\begin{aligned}
&\Rightarrow 4\ddot{x} + \alpha f' = 0 \\
&\Rightarrow 4(\ddot{x} - \ddot{x}_d) + 4k'_v(\dot{x} - \dot{x}_d) + 4k'_p(x - x_d) = 0 \\
&\Rightarrow 4\ddot{e} + 4k'_v\dot{e} + 4k'_pe = 0
\end{aligned}$$

The coefficient of the  $e$  term is the closed loop stiffness, so:  $k'_p = 9$ .

In order to have critical damping, we need to have  $\xi = 1$ . Using Equation 7.12 we see that the coefficient of  $\dot{e}$  must be:

$$k'_v = 2\sqrt{k'_p} = 6$$

Thus, the control is

$$\begin{aligned}
f &= \alpha f' + \beta \\
\alpha &= 4 \\
\beta &= 30\text{sign}(\dot{x}) + 25x \\
f' &= \ddot{x}_d - 6(\dot{x} - \dot{x}_d) - 9(x - x_d)
\end{aligned}$$

- (d) **Given a disturbance force  $f_{dist} = 4$ , what is the steady-state ( $\ddot{e} = \dot{e} = 0$ ) error of the system in part (c)?**

We can analyze the error by observing the error in the unit-mass system. With a disturbance force added, the system's equation of motion becomes

$$4\ddot{x} + 30\text{sign}(\dot{x}) + 25x = f + f_{dist}$$

To linearize the system, we apply a force of the same form as before:

$$\begin{aligned}
f + f_{dist} &= 4f' + 30\text{sign}(\dot{x}) + 25x + f_{dist} \\
&= 4\left(f' + \frac{f_{dist}}{4}\right) + 30\text{sign}(\dot{x}) + 25x
\end{aligned}$$

This yields a unit-mass system as before, but now it has a disturbance force of  $f_{dist}/4$ , so the unit-mass system now looks like

$$\ddot{x} = f' + \frac{f_{dist}}{4}$$

With the control from before, we get a unit-mass closed-loop system of

$$\ddot{e} + 6\dot{e} + 9e = \frac{f_{dist}}{4}$$

For the steady state, when  $\ddot{x} = \dot{x} = 0$ , we get

$$9e = \frac{f_{dist}}{4}$$

So, the steady state error is given by

$$e = \frac{f_{dist}}{4 \cdot 9} = \frac{4}{36} \approx 0.111$$

2. **For a certain RR manipulator, the equations of motion are given by**

$$\begin{bmatrix} 4 + c_2 & 1 + c_2 \\ 1 + c_2 & 1 \end{bmatrix} \begin{bmatrix} \ddot{\theta}_1 \\ \ddot{\theta}_2 \end{bmatrix} + \begin{bmatrix} -s_2(\dot{\theta}_2^2 + 2\dot{\theta}_1\dot{\theta}_2) \\ s_2\dot{\theta}_1^2 \end{bmatrix} = \begin{bmatrix} \tau_1 \\ \tau_2 \end{bmatrix}$$

- (a) **Assume that joint 2 is locked at some value  $\theta_2$  using brakes and joint 1 is controlled with a PD controller,  $\tau_1 = -40\dot{\theta}_1 - 400(\theta_1 - \theta_{1d})$ . What is the minimum and maximum inertia perceived at joint 1 as we vary  $\theta_2$ ? What are the corresponding closed-loop frequencies?**

For joint 2 locked ( $\ddot{\theta}_2 = \dot{\theta}_2 = 0$ ), the equation of motion for joint 1 is:

$$(4 + c_2)\ddot{\theta}_1 = \tau_1$$

The inertia seen at joint 1 is the coefficient of the  $\ddot{\theta}_1$  term,  $(4 + c_2)$ . So, this inertia achieves its maximum and minimum values at  $\theta_2 = 0$  and  $\theta_2 = 180^\circ$ :

$$m_{max} = 5, \quad m_{min} = 3$$

The closed-loop equation for joint 1 is

$$(4 + c_2)\ddot{\theta}_1 + 40\dot{\theta}_1 + 400(\theta_1 - \theta_{1d}) = 0$$

To get an expression for closed loop frequency, we compare our closed loop equation with the generic system of Equation 7.9 ( $m\ddot{x} + b\dot{x} + kx = 0$ ).

The closed loop frequency is then given by:

$$\omega = \sqrt{\frac{k}{m}} = \sqrt{\frac{400}{(4 + c_2)}}$$

So, we have

$$\begin{aligned} m = m_{max} &\Rightarrow \omega_{min} = \frac{20}{\sqrt{5}} \\ m = m_{min} &\Rightarrow \omega_{max} = \frac{20}{\sqrt{3}} \end{aligned}$$

- (b) **Still assuming that joint 2 is locked, at what values of  $\theta_2$  do the minimum and maximum damping ratios occur? What are the minimum and maximum damping ratios?**

To get an expression for damping ratio, we once again compare our closed loop equation with the generic system of Equation 7.9 ( $m\ddot{x} + b\dot{x} + kx = 0$ ). In this case, using Equation 7.12 the closed-loop damping ratio is given by:

$$\xi = \frac{b}{2\sqrt{km}} = \frac{40}{2\sqrt{400m}} = \frac{1}{\sqrt{m}} = \frac{1}{\sqrt{4 + c_2}}$$

So, the minimum and maximum values of  $\xi$  occur at  $\theta_2 = 0$  and  $\theta_2 = 180^\circ$ :

$$\begin{aligned} \xi_{min} &= \frac{1}{\sqrt{m_{max}}} = \frac{1}{\sqrt{5}} \\ \xi_{max} &= \frac{1}{\sqrt{m_{min}}} = \frac{1}{\sqrt{3}} \end{aligned}$$

- (c) **Now assume that both joints are free to move, and that this system is controlled by a partitioned PD controller,  $\tau = \alpha\tau' + \beta$ . Design a partitioned, trajectory-following controller (one that tracks a desired position, velocity and acceleration) which will provide a closed-loop frequency of 10 rad/sec**

on joint 1 and 20 rad/sec on joint 2 and be critically damped over the entire workspace. That is, let

$$\tau' = \ddot{\theta}_d - \begin{bmatrix} k'_{v1} & 0 \\ 0 & k'_{v2} \end{bmatrix} (\dot{\theta} - \dot{\theta}_d) - \begin{bmatrix} k'_{p1} & 0 \\ 0 & k'_{p2} \end{bmatrix} (\theta - \theta_d),$$

then find the matrices  $\alpha$  and  $\beta$  and the vector  $\tau$ , along with the necessary gains  $k'_{v_i}$  and  $k'_{p_i}$ .

The equations of motion are of the form

$$M(\theta)\ddot{\theta} + V(\dot{\theta}, \theta) = \tau$$

to which we apply a vector of torques  $\tau$  of the form

$$\tau = \alpha\tau' + \beta$$

To make this look like a unit-mass system, we let

$$\alpha = M(\theta), \quad \beta = V(\dot{\theta}, \theta)$$

which gives the unit-mass system

$$\ddot{\theta} = \tau'$$

To this system, we apply the control

$$\tau' = \ddot{\theta}_d - \begin{bmatrix} k'_{v1} & 0 \\ 0 & k'_{v2} \end{bmatrix} (\dot{\theta} - \dot{\theta}_d) - \begin{bmatrix} k'_{p1} & 0 \\ 0 & k'_{p2} \end{bmatrix} (\theta - \theta_d),$$

This yields two closed-loop equations

$$\begin{aligned} \ddot{e}_1 + k'_{v1}\dot{e}_1 + k'_{p1}e_1 &= 0 \\ \ddot{e}_2 + k'_{v2}\dot{e}_2 + k'_{p2}e_2 &= 0 \end{aligned}$$

where  $e_i$  is the error at joint  $i$ ,  $e_i = (\theta_i - \theta_{id})$ . Now, we need to choose  $k'_{v_i}$  and  $k'_{p_i}$  to achieve critical damping, and to achieve our desired closed-loop frequencies. For a unit-mass system, we choose

$$\begin{aligned} k'_{p_i} &= \omega_i^2 \\ k'_{v_i} &= 2\xi_i\omega_i \end{aligned}$$

So, we get

$$\begin{aligned} k'_{p1} &= 100, \quad k'_{v1} = 20 \\ k'_{p2} &= 400, \quad k'_{v2} = 40 \end{aligned}$$

- (d) **If  $\theta_2 = 180^\circ$ , what is the steady-state error vector for a given disturbance torque,  $\tau_{dist} = [2 \ 4]^T$ ?**

The controlled system, with a disturbance torque  $\tau_{dist}$  is

$$M(\theta)\ddot{\theta} + V(\dot{\theta}, \theta) = \tau + \tau_{dist}$$

Substituting in our form for  $\tau = \alpha\tau' + \beta$  yields

$$M(\theta)\ddot{\theta} - M(\theta)\tau' = \tau_{dist}$$

This has the form

$$M(\theta) \left[ \ddot{\mathbf{e}} + K'_v \dot{\mathbf{e}} + K'_p \mathbf{e} \right] = \tau_{dist}$$

where  $\mathbf{e}$  is the error vector  $\mathbf{e} = \theta - \theta_d$ , and  $K'_v$  and  $K'_p$  are the matrices given by

$$K'_v = \begin{bmatrix} k'_{v1} & 0 \\ 0 & k'_{v2} \end{bmatrix}, \quad K'_p = \begin{bmatrix} k'_{p1} & 0 \\ 0 & k'_{p2} \end{bmatrix}$$

In the steady state ( $\ddot{\mathbf{e}} = \dot{\mathbf{e}} = \mathbf{0}$ ), the equation is

$$M(\theta) K'_p \mathbf{e} = \tau_{dist}$$

which means that the steady state error is

$$\mathbf{e} = (M(\theta) K'_p)^{-1} \tau_{dist}$$

For our values, this is:

$$\mathbf{e} = \left( \begin{bmatrix} 3 & 0 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} 100 & 0 \\ 0 & 400 \end{bmatrix} \right)^{-1} \begin{bmatrix} 2 \\ 4 \end{bmatrix} = \begin{bmatrix} 300 & 0 \\ 0 & 400 \end{bmatrix}^{-1} \begin{bmatrix} 2 \\ 4 \end{bmatrix} = \begin{bmatrix} \frac{1}{150} \\ \frac{1}{100} \end{bmatrix}$$